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CONF-9608132--7

Title:

PROCYON HIGH EXPLOSIVE PULSED POWER EXPERIMENTS

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Author(s):

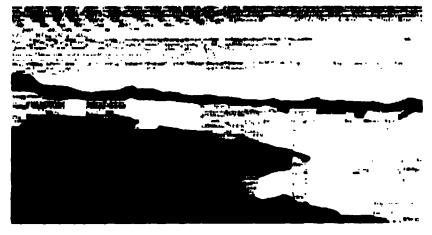
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Submitted to:

7th International Conference on Megagauss Megaetic Field Generation and Related Topica August 5-10, 1996 Sarov (Arzamas-16), RUSSIA

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# PROCYON HIGH EXPLOSIVE PULSED POWER EXPERIMENTS

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University of California, Los Alames National Laboratory, Los Alames, NM, USA

#### Introduction

Procyon is a two-stage explosive pulsed-power system, consisting of a MK-IX helical generator' and an explosively formed tuse' (EFF) opening switch. A complete assembly including load and diagnostics is shown if Fig. 1. The system was originally developed for the purpose of powering plasma z-pinch experiments and, in its original concept, was coupled to the plasma z-pinch load through a third pulsed power stage, a pleama flow switch (PFS)\*. We have performed plasma z-pinch experiments both with and without a PFS, and we have now conducted our first heavy liner experiment. In this paper, we will summarize the results obtained to date with the system, and briefly discuss future applications.

## PFB Tests

Our original design goals for I rooyon were driven by the belief that to obtain the radiation temperature desired, a sub-microsecond plasma implesion would be required. From previous experiments, we know that we could make an EFF that would satisfy this need, but transmission line voltages appeared excessive, and our baseline design included a slower opening EFF along with a plasma flow switch for an intermediate stage oponing switch between the CFF and the implesion. Several tests were performed in this configuration, and we achieved significant results on two of these. We reported the results of our initial static load test in the 1993 ICEE Pulsed Power Conference Proceedings\*. Currents measured on our most successful z-pinch lond test are given in Fig. 2. On this test we delivered 15 MA from the storage inductor to the PFS, and the pinch occurred at over 14 MA. The PFS did not employ a conventional 1/r gun plasma mass distribution, but varied as 1/r, causing the switch to open while plasms was still in the barrel, as in Fig. 3a. Only minimal radiation was detected by external diagnostics, hacsuse of the mass of PFS plasms that filled the detector viewing port at the critical time, as shown in Fig. 3b. From an energy analysis using the currents shown, we conclude that -1MJ was dissipated on a time scale of interest for radiation . Dased on experience from other experiments, we believe that we could have generated as much as 750 KJ of useful radiation. Experiments using the radiation pulse would be located in a position not attended by the PFB plasma, and the occluded radiation probably done not nose a significant problem.

## Direct Drive Experiments

We concluded three Progres experiments in which we switched current to a plasma impleation directly with the voltage produced by the EFF. Each of those tests produced useful rediation, and our both radiation results were obtained this way. On our nighest fluence test, we dissipated 2 MJ from the circuit, and measured 1.6 MJ radiation at 60 eV. Currents from this test are shown in Fig. 4, and the x-ray pulse shape is shown in Fig. 5. Who a full width at half maximum of 200 is we inter an average power for the radiation of 30.



Figure 1. Procyce assembly mady to test on the firing pad. The MK IX generator is on the far left, the strange inductor and opining switch are in the central section, and the implement and diagnostics chamber on the right.

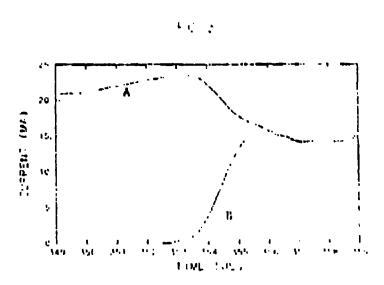
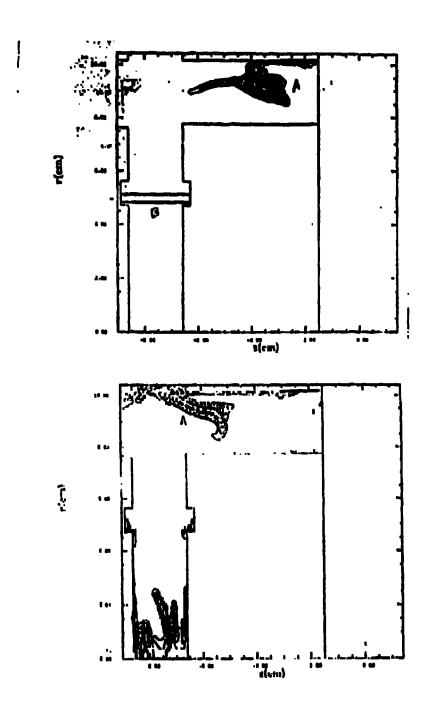


Figure 2. Storage inductor (A) and PFS (B) currents for test with implication load. The first dip in the PFS current is due to the PFS switching and the second is due to the planma z-plinch.



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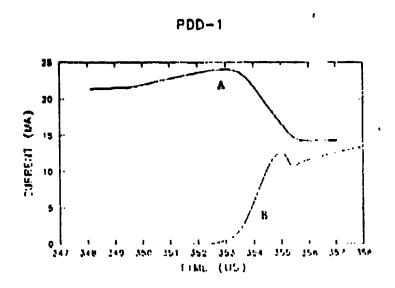
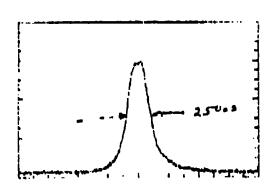


Figure 4. Storage inductor (A) and load (B) currents for test producing 1 fi MJ radiation. The dip at -355  $\mu s$  indicates pinch occurring.



) igure 5. Kimfoll filtered x ray diode signal from 1.6 MJ radiation toxit. If all width at half maximum is -250 ns indicating an average radiation power of -61W.

TW. One of our three tests employed a contoured electrode by extending a conical protrusion approximately half way across the 2-cm gap, starting at approximately half the initial plasms radius. The primary purpose for the contour was to encourage a high-velocity jet to be emitted adaily from the pinch region. The jet was impacted on a high-density target to produce radiation. A short-duration (~50ns), high-temperature (~80eV) radiation pulse was measured from the target region, and in addition, the contour seemed to reduce or eliminate adai instabilities. Figure 6 shows visible framing camera records from one test with parallel electrodes to compare with the shaped electrode test. Although the instabilities shown for the parallel electrode test are among the worst observed, the difference shown is dramatic. A more thorough discussion of these data are given in another paper in this conference?. We generally consider the instabilities harmful, and the radiation fluence from the parallel electrode shot shown was about one half of that on our best shot. However, recent analysis of the radiation produced by the highly unstable pinch indicates that this may have been the nighest temperature, ~97 eV, of any we produced? Although we have uncovered a rich ground for further exploration, changing programmatic goals preclude our pursuing them further, and we present those exploratory results for others who may have the opportunity.

#### Solid Liner Tests

The emphasis in our work has now turned to producing high pressure shock waves by impacting solid density liners onto targets of interest. The Procyon system also allows us to perform such experiments at high energy (and with high di/dt), and we have performed one proliminary solid liner experiment. Recause the solid finor experiments do not produce large quantities of radiation during their implosion phase, the radiation traffics needed to protect the vacuum dielectric interface on plasma implesion tests are not needed. This allows us to implement a lower inductance load, as illustrated in Fig. 7. With the reduced inductance of the load shown, we expect to be able to deliver almost 20 MA to solid liner loads. Figure 8 shows the currents measured in our first experiment. A partial failure during the operation of the MK-IX generator resulted in a storage inductor current of only 19.1 MA (as opposed to the 21-22.5 MA that has been a very reliable value for this system). This, in turn, led to a load current of only 16 MA. The liner chosen for this test was 12 mg of aluminum, and the reduced current profile led to a slower implication than expected. Since this was a preliminary experiment, we had no target for the liner to impact, and diagnostics were limited. However, as can be seen from the waveform, the ther achieved significant implesion velocity by ~360 us, and the implesion profile is consistent with the imposed waveform. Apart from verifying that our low inductance load coupling was satisfactory, possibly the most significant result from the test was provided by the visible framing camera that viewed the outside of the liner. By itself, the record is confusing and difficult to explain. The outside of the liner appears to blow material off the surface. There is a striated effect along the side of the cylinder, that could correlate with machining imperfections on the liner, although the spacing of the strictions does not correlate in any why with machining tolorance. In addition, there are apparently random effects that do not correlate with any observed imperications. We have recently performed an experiment on the Pagasus facility that allows up to have increased confidence in this interpretation. The same phenomena am observed on this test, although reduced in magnitude and with fewer random effects. Further study is required, but since di/dt on the Procyon test in <10 MA/us , while the dI/dI on Pegasus in >1.5 MA/us, we may be observing a limit on liner surface quality that must be dealt with as we pursue blaher currents and rate of delivery.

### **Future Tests**

Stability of imploring liners plays a very important role in the future success of our afforts. In order to drive high-mass known to high implesion velocities with reasonable efficiency, high convergence ratios must be achieved. We are currently conducting experiments on the Pegasus capsolier bank, which has a restrict of 8 and a current of 4-12 MA. We have also conducted a preliminary liner stability test with the Procyon system to demonstrate that our low inductance power feed will function adequately for the liner loads. After gathering chibitily data on Pegasus for a parameter of interest, our intention is to observe the same parameters using the Procyon system that one give us higher hand divid. In this way, we can obtain very useful information about how various parameters vary as we approach levels that we will achieve on the Allas capacitor bank. One of our entirest investigations will be to see if a third layer of point material will land abilities to an otherwise medical liner. This was the capacity is less than autoquates.

The was the physics mad for our first Procyce theor test. Since the current schools we have then enterpolarly and sense parties and the last ref me me made it is a great along, is ever a first picture.

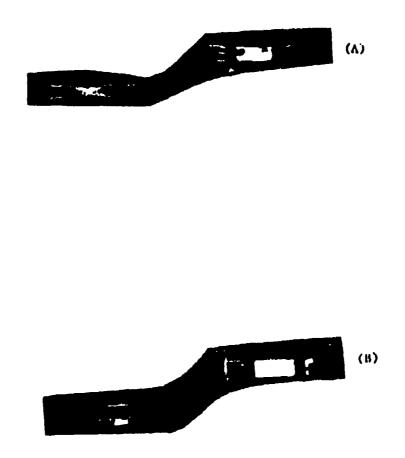
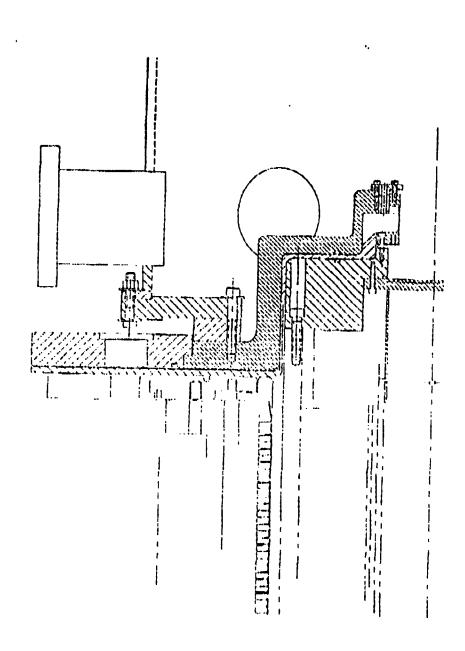


Figure 6. Framing camera records from (a) parallel electrode test and (b) contoured electrode test. A large instability is seen in (a) in the frame before pinch, while such an instability is not seen with the contoured electrode.



f-tours 7. Hoduced inductance load configuration for Procyon liner touts. Configuration allows complete radial access to the load, and exall access from one and - With this load, we should be able to deliver almost 20 MA to a liner load.

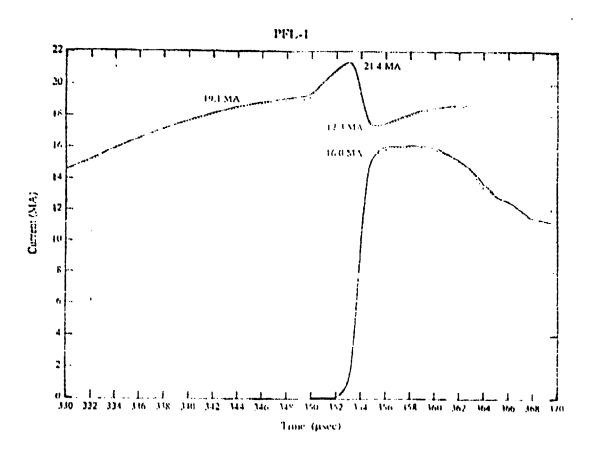


Figure 8. Storage (A) and load (B)) currents for our first Procyon Liner Experiment. Implosion begins to affect the waveform at  $\sim 360~\mu s$ .

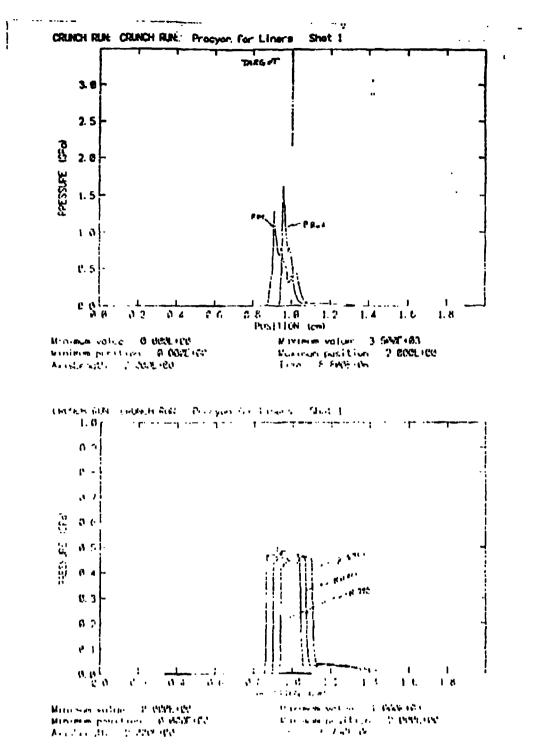


Figure 9. Pressure profiles calculated for Procyon liner lests, (a) Shows the high pressure space generaled by a thin platinum layer on an alaminum liner. (b) Shows the profile from an arangonic form. The target outlond is platinum in both cases.

liner at the right time to observe its condition when it was mostly melted. However, with diagnostic improvements that we are making based on the experience of our first test, and with Pegasus tests in addition, we are confident that we will be able to achieve this goal on subsequent tests. Other tests will follow as we learn more of the parameters of interest. In addition, we can begin developing diagnostic capability on Procyon that will be important for Atlas experiments. We have made computer simulations of the kind of pressures that can be achieved by driving liners into targets of interest using the Procyon system. Figure 9 shows two plots of pressure profiles that can be achieved in a Procyon experiment. With a layer of Platinum on the inner surface of an aluminum liner, pressures of -15 MB can be achieved for short times, or pressures of -5 MB can be maintained for considerably longer times. Using such liners, we can develop the ability to diagnose the pressures produced in such a way, and further to perform experiments in this environment.

#### Conclusions

We have demonstrated that Procyon is a reliable system for PFS, plasma implosion or heavy liner experiments. Procyon develops 18 MJ magnetic energy, and performance parameters allow experiments to be conducted in the range where multiple megajoules are delivered to a load. We have recently been performing experiments in which a castable explosive is used with our axial detonation system, which can save considerable explosive expense on each shot. In preparation for higher energy tests, both with fixed machines and even larger explosive pulsed power devices, we hope to study liner physics issues in future test series, and perhaps commence preliminary high-pressure physics tests.

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